

# **GLAST Large Area Telescope**Calorimeter Subsystem 6.0 Mechanical Subsystem

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#### **CDR Outline**

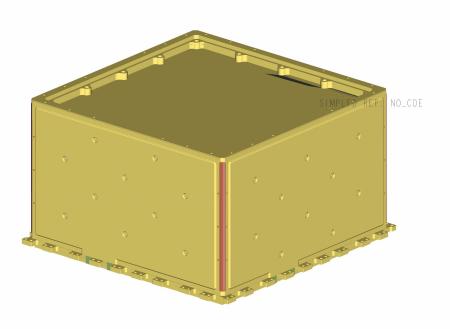
- □ CAL Mechanical Subsystem
  - Subsystem Overview

**Paul Dizon** 

- Mechanical Design and Analysis Oscar Ferreira
- Thermal Design and Analysis Pierre Prat
- Manufacturing of Flight Modules Paul Dizon
- Summary Paul Dizon



# **CAL Design Overview**



Key Parameters					
Dimensions Max Length & Width Min Length & Width		2			
Height		221.8 mm			
Mass Structure Electronics CDE Total		9.6 kg 1.7 kg 75.7 kg 87.0 kg			
First Mode Frequency		>150 Hz			
Max Deflection	Static Load Random Vibration	0.3 mm 0.3 mm RMS			





## **CAL Design Overview**

- Pre-Electronics Module Assembly
  - PEM Mechanical Structure
    - Carbon Fiber Structure
    - Aluminum Base Plate and Top Frame
  - Crystal Detector Elements (CDE) 96 Units
  - Close-Out Plates
- □ CAL Module Assembly
  - PEM Assembly
  - AFEE Cards
  - Side Panels
  - TEM Stand-Off
  - TEM Harness and Associated Brackets





#### **CAL Design Overview**

#### Engineering Challenges

- Maximize Packaging Volume for AFEE While Minimizing Effective Detector Gaps Between Adjacent CAL Modules
- Ensure Structural Stiffness of the CAL Modules without Relying on the Mechanical Properties of the Csl Crystal Logs
- Secure CsI Crystal Logs for Launch Loads without Constraining them During Thermal Expansion





## **Derived Requirements**

Document Title	Document Number	Status
Mechanical Systems Interface Definition Drawing, CAL - LAT Inter	fat <b>A</b> T-DS-00233	Released
LAT Contamination Control Plan	LAT-MD-00404-02	Released
Interface Control Document between the Calorimeter Subsystem LAT Instrument	and LAT-SS-00238-04	Released
CAL Mechanical Structures Specification	LAT-SS-00241-03	Released
LAT Environmental Specification	LAT-SS-00778-01-D8	In signoff





## **Compliance Matrix**

Parameter	Parameter Requirement Compliance C		
Physical Interfaces & Clearance - Dimensions & Tolerances - Surface Condition of Tabs - Static Stay-Clear Dimensions - Dynamic Stay-Clear Dimensions - Lateral - Vertical  Base Plate Requirements - Stiffness - CTE	per IDD - LAT-DS-00233  per LAT-SS-00238 0.50 mm MAX 0.50 mm MAX  per LAT-SS-00238 Equivalent to 8 mm Al Plate CTE = 21-25E-6 m/m/deq C	YES YES YES YES YES	Measured on EM Measured on EM Measured on EM  Verified on VM2 (<0.3 mm)  Verified on VM2 (<0.3 mm)  Equivalent to 8 mm Thickness (Plus Margin)  Aluminum Plate
Mass Properties - Mass - Center of Gravity  Structural Load Environment	per LAT-SS-00238 90 kg Max Lateral Offset +/-10 mm Max Z CG Position = 116 mm  per LAT-SS-00238/LAT-SS-00778	YES YES YES	Measured on EM (approx 86 kg) Analysis Shows within +/- 1 mm Analysis Shows Zcg =87.5 mm from CAL-Grid Interface  Verified on VM2. To be Re-verified on EM (Analysis Shows
- Static-Equivalent Accelerations	Lateral = $+/-6.0$ g Axial = $+6.8/-1.8$ g		Positive Margins) Verified on VM2. To be Re-verified on EM (Analysis Shows Positive Margins)
- Random Vibration		YES	Verified on VM2. To be Re-verified on EM (Analysis Shows Positive Margins)
- Interface Limit Loads	4373 N max across tab	YES	To be Verified on EM (Analysis Shows Positive Margin)
- Interface Distortion Limit Loads	0.25 mm max distortion	YES	Analysis Shows Positive Margin
- TEM Interface Loads	Tension = 3750 N Compression = 2625 N Shear = 1288 N Bending Moment = 19.3 Nm	YES	To be Verified on EM (Analysis Shows Positive Margin)



## **Compliance Matrix - Continued**

Parameter	Parameter Requirement		Comments
Thermal Environment  - CAL-Grid Conductance  - CAL-TEM/PS Contact Conductance  - Operating Temperature  - Survival Temperature	per LAT-SS-00238/LAT-SS-00778 >0.03 W/cm² deg C  0.1 W/dec C +25/-15 deg C +50/-30 deg C	YES YES YES YES	Analysis uses 0.25 W/cm2 deg C. Analysis to be updated following TVAC Testing Analysis used 0.1 W/deg C. Analysis to be updated following TVAC Testing Analysis shows CAL Meets Temperature Requirements Analysis shows CAL Meets Temperature Requirements
First Natural Frequency	<b>per LAT-SS-00238</b> > 50 Hz	YES	Verified on VM2 and by Analysis (>150 Hz). To be Reverified on EM
Venting	per LAT-SS-00238/LAT-SS-00778	YES	Analysis Shows Positive Margin
Preserve Safety of CDEs	No Light Yield Change	YES	Verified on VM2
Contamination/Particulates	per LAT-SS-00238	YES	All Particulates Generated from Fracture-Sensitive Materials will be Contained within the Stay-Clear Volume of the CAL
	per LAT-MD-00404	YES	All materials approved for flight Structural cleanliness addressed during assembly Bake out of all composite and polymer materials





## **Changes Since PDR**

- AFEE Card Packaging Volume Resizing
  - AFEE Card Redesign Resulted in Required Increase in Packaging Volume. Increased Packaging Volume by:
    - Shortening Length of CDE
    - Decreasing the Distance Between Opposing Close-Out Plates by Modifying Base Plate and Top Frame Interface Dimensions
- Close-Out Plate Redesign
  - Incorporated Stiffeners to Minimize Deflection
  - Openings for Electrical Interconnects between CDE PDA and AFEE Redesigned due to Removal Flex Cable from Design:
    - Geometry of Openings for Electrical Interconnects Modified to Accommodate New Design
    - Position of Openings for Electrical Interconnects
       Changed to Reflect New Position of AFEE Components





## **Changes Since PDR - Continued**

- Base Plate Modifications Requested by SLAC
  - Modifications were Required to Reduce Stiffness in order to Decrease Bolt Loads at the Tab-LAT Grid Interface
    - Increase Tab Length
    - Decrease Tab Thickness
  - Maximize Friction Characteristics of the Bolted Joint
    - Reduce Outer Radius of Tab Corners
    - Remove Chamfer of Upper and Lower Surface of Tab
    - Remove Surface Treatment on Upper and Lower Surface of Tab
- Additional Base Plate Modifications
  - TEM Interface
    - Flex-Mount changed to Hard-Mount Interface
  - TEM Cable Bracket Interface





## **Changes Since PDR - Continued**

- Improvement of Cure Process for Composite Structure
  - Improved Curing Process Has Better Pressure Control
    - Vacuum Bagging
    - Autoclave
  - Tooling Redesigned to Accommodate New Curing Process
- CDE Interface
  - Bumper Frame Redesign
    - Added End Caps to Eliminate PDA Clearance Issues with the Original Bumper Design
- Assembly Tooling and Procedures
  - Minor Changes due to Change from Flex Cable to Twisted Wire Pairs for PDA Electrical Interconnects





#### **Status of PDR RFAs**

□ No RFAs Assigned at the PDR





#### **Schedule**

#### □ Top Level Milestones

Completed EM PEM Assembly 10 Feb 2003

Complete EM CAL Module Assembly 31 Mar 2003

Start EM Structural Environmental Testing
 Apr 2003

- Production of SM CAL Test Structure 09 Apr 2003

- Start EM Thermal Vacuum Testing 06 May 2003

Production of SFM CAL Test Structure 30 June 2003

Delivery of Flight PEM Structures 24 Jul 2003 thru
 Mar 2004



## **CAL Subsystem Mass**

#### **Mechanical Structure Mass for Each CAL Module**

1	Part or Dwg No. AT-DS-01115-01	Component Name/Desc. CSI Crystal PIN Photodiode Twisted Wire Pair Optical Bond Material Optical Reflective Wrap	<b>Quantity</b> 96 192 192	Mass (Kg) 0.773	Mass Estimation Method (PARA, CALC, or MEAS)	Total (Kg)	Development Class (1, 2, or 3)	Calculated Contingency Recommendation (ANSI/AIAA-G-020-	X-Axis Center of Mass	Y-Axis Center of Mass	Z-Axis Center of
1		Csl Crystal PIN Photodiode Twisted Wire Pair Optical Bond Material	96 192	0.773		Total (Kg)	2 05 21				
2 3 4 4 4 5 5 6 7 7 7 8 9 10 10 11 11 12 12 14 15 15 16 17 17 18 18 18 18 18 18 18 18 18 18 18 18 18	AT-DS-01115-01	PIN Photodiode Twisted Wire Pair Optical Bond Material	192				2, UF 3)	1992)	(mm)	(mm)	Mass (mm)
3 4 4 5 6 7 7 7 8 9 10 10 11 11 12 13 14 15		Twisted Wire Pair Optical Bond Material			MEAS	74.241	3	2%		-	-146.37
4 4 5 6 7 7 8 8 9 10 L/11 L/12 L/13 L/14 15 L/15 L/16 15 L/16 L/16 15 L/16 15 L/16 15 L/16 L/16 15 L/16 L/16 L/16 L/16 L/16 L/16 L/16 L/16		Optical Bond Material	192	0.0016	MEAS	0.307	3	2%	-	-	-146.37
4 5 6 7 7 7 8 8 9 10 L/11 L/12 L/13 L/14 15 L/15 L/15 L/15 L/15 L/15 L/15 L/15 L				0.0001	CALC	0.019		20%	-	-	-146.37
5 6 7 7 8 9 10 L/ 11 L/ 12 L/ 13 L/ 14		Ontical Reflective Wran	192	0.0002	MEAS	0.038		2%	-	-	-146.37
6 7 7 8 9 10 L/11 L/12 L/13 L/14 15 L/15 L/16			96	0.0033	MEAS	0.317	3	2%	-	-	-146.37
7 7 8 9 10 L/1 11 L/1 12 L/1 13 L/1 14 15 L/1		End Cap	192	0.0006	MEAS	0.115	2	20%	-	-	-146.37
7 8 9 10 L/ 10 L/ 11 L/ 12 L/ 13 L/ 14								-			
8 9 10 L/1 11 L/1 12 L/1 13 L/1 14 15 L/1		Printed Circuit Board (AFEE)	4	0.360	CALC	1.440		20%	-	-	-134.16
9 10 L/ 10 L/ 11 L/ 12 L/ 13 L/ 14		AFEE-TEM Cable	4	0.050	CALC	0.200		25%	-	-	-256.20
10 L/ 10 L/ 11 L/ 12 L/ 13 L/ 14 L/ 15 L/		AFEE-TEM Cable Bracket	4	0.005	PARA	0.020	1	25%	-	-	-256.20
10 L/ 11 L/ 12 L/ 13 L/ 14 L/ 15 L/								-			
11 L/ 12 L/ 13 L/ 14 L/ 15 L/		Composite Structure	1	2.651	MEAS	2.651		20%	-	-	-147.20
12 L/ 13 L/ 14 L/ 15 L/		EMI Shield	1	0.029	MEAS	0.029		2%	-	-	-45.40
13 L/ 14 15 L/		Insert, Side	40	0.001	MEAS	0.057	3	2%	-	-	-147.20
14 15 L/		Insert, Top Composite	16	0.001	MEAS	0.023		2%	-	-	-147.20
15 L/	AT-DS-00929-01	Insert, Bottom Composite	25	0.005	MEAS	0.114	3	2%	-	-	-147.20
								-			
	AT-DS-00917-01	Top Frame	1	0.625	MEAS	0.625	3	2%	-	-	-36.45
		Base Plate	1	3.190	PARA	3.190	3	2%		-	-238.17
		Close-Out Plate X Assembly	2	0.330	MEAS	0.660	3	2%	-	-	-133.23
	AT-DS-00921-01	Close-Out Plate Y Assembly	2	0.330	MEAS	0.660		2%		-	-133.23
19 L/	AT-DS-00923-01	Side Panel X	2	0.146	MEAS	0.292	3	2%	-	-	-132.78
	AT-DS-00924-01	Side Panel Y	2	0.146	MEAS	0.292	3	2%	-	-	-132.78
21								-			
22 L/	AT-DS-00922-01	Nut	40	0.0016	MEAS	0.064	3	2%	-	-	-147.20
23								-	-	-	
		Bumper (Elastomer)	192	0.0004	MEAS	0.077	2	20%	-	-	-146.37
		Bumper Frame (Delrin)	192	0.0004	MEAS	0.077	2	20%	1	-	-146.37
26 L/	AT-DS-00926-01	Spacer (Omit from Current Design)			PARA			-			
27		Elastomeric Cords	384	0.0002	MEAS	0.077	2	20%		-	-146.37
28											
29		Fasteners	1	0.300	MEAS	0.300	3	2%	-	-	-146.37
30								-			
31		Miscellaneous	1	0.100	PARA	0.100	1	25%	-	-	-146.37
32								-			
			Total 1	Т	Total Mass otal PARA Mass otal CALC Mass otal MEAS Mass	85.985 3.310 1.659 81.016 85.985	3.8% 1.9% <u>94.2%</u>	3% Total Mass Contingency	0.00 Cen	0.00 hter of Mass(x	-148.75 x,y,z)
			Total for		Total Mass otal PARA Mass otal CALC Mass	1375.759 52.960 26.547	3.8%	3% Total Mass Contingency		0.00 Iter of Mass(x For this Modul	





## **Mechanical Design Status**

Mec	hani	ical	Design
1-100		CGI	203.9

- Base Plate	100	<b>%Complete</b>
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- Top Frame	100	<b>%Complete</b>
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- EMI Shield 100 %Complete
- Composite Structure 100 %Complete
- Composite Structure Flt Tooling 75 % Complete
- Bumper Frame/End-Cap Design 100 %Complete
- Close-Out Plate Assembly
- Side Panels

□ Drawings 90 % Complete

□ Specifications/Procedures 75 % Complete



**%Complete** 

95 % Complete

100



# **CAL Module Fabrication Drawings**

DRAWING	DRAWING NUMBER
CALORIMETER ASSEMBLIES  EM Pre-Electronics Module (PEM) Assembly PEM Mechanical Structure Shipping Configuration PEM Mechanical Structure	LAT-DS-00916-02 LAT-DS-01224-01 LAT-DS-01228-01 LAT-DS-01231-01
MECHANICAL STRUCTURE PIECE PARTS  Top Frame Composite Structure Base Plate Close Out Plate X Close Out Plate Y Nut Side Panel X Side Panel Y Bumper Frame Insert, Side Insert, Top Composite Insert, Bottom Composite EMI Shield	LAT-DS-00917-01 LAT-DS-00918-02 LAT-DS-00919-04 LAT-DS-00920-03 LAT-DS-00921-03 LAT-DS-00922-01 LAT-DS-00923-02 LAT-DS-00924-02 LAT-DS-00925-02 LAT-DS-00927-01 LAT-DS-00928-01 LAT-DS-00929-01 LAT-DS-01234-01

